QUASISTATIC RESPONSE OF AN INFINITE HOLLOW VISCOELASTIC CYLINDER UNDER TIME VARYING INTERNAL PRESSURE

R. N. KANANGO & N. PATNAIK

University College of Engineering, Sambalpur

(Received 8 October 1973; revised 18 July 1974)

The quasistatic response of an infinite hollow viscoelasti evidence in terms of radial displacement, strain and stress components is derived from the fundamentals corresponding to a time varying internal pressure by using Fourier transform method and as a specific example, the problem of an infinite hollow cylinder, made of ebsnite and subject to constant internal pressure, has been solved analytically.

Analytical solutions for the quasistatic problems of viscoelastic cylinders have been presented earlier by Lee¹, Bland² and Christensen³ using Laplace transform method. The solutions for these problems are all based on the elastic-viscoelastic correspondence principle.

The present paper deals with the solution expressed in the form of displacement, strain and stress components within an infinite hollow viscoelastic cylinder subject to time varying internal pressure. The Fourier transform method which has been used here to obtain the solution, is considered more convenient for obtaining the quantitative solution of the problems concerning viscoelastic materials with the properties expressed as complex function of frequency.

FIELD EQUATIONS

In polar cylindrical coordinates r, θ , z with axial symmetry the nontrival elastic equation of motion is given by

$$\frac{\partial^2 u}{\partial r^2} + \frac{1}{r} \frac{\partial u}{\partial r} - \frac{u}{r^2} = 0$$
 (1)

where, u = u(r, t) is the radial displacement. Using Fourier transform method, the equation is converted to

$$\frac{\partial^2 \bar{u}}{\partial r^2} + \frac{1}{r} \frac{\partial \bar{u}}{\partial r} - \frac{\bar{u}}{r^2} = 0$$
 (2)

where $\vec{u} = \vec{u}$ (r, w), w being transform parameter. Strain-displacement and stress-strain relation are given by

$$\tilde{\epsilon}_{rr} = \frac{\partial \bar{u}}{\partial r}, \epsilon_{\theta\theta} = \frac{\bar{u}}{r}, \tilde{\epsilon}_{zz} = 0$$
 (3)

$$\bar{\sigma}_{rr} = 2\bar{\mu} \left(\frac{\partial \bar{u}}{\partial r} + \frac{\bar{\nu}}{1 - 2\bar{\nu}} \cdot \bar{e} \right)$$
(4)

$$\overline{\sigma}_{\theta\theta}^{i\theta} = 2\overline{\mu} \left(\frac{\overline{u}}{r} + \frac{\overline{\nu}}{1-2\overline{\nu}} \cdot \overline{e} \right)$$
(5)

$$\bar{\sigma}_{zz} = 2\bar{\mu} \left(\frac{\bar{\nu}}{1-2\bar{\nu}} \right). \bar{e}$$
(6)

where

(?)

 $\mathbf{p}^{(i)}$

 $\bar{\mu} = \text{Transformed shear modulus} = \mu^*$ (iw)

 $\tilde{\nu} =$ Transformed Poisson's ratio

antia ngadata Tangatan tan

 $\bar{e} = \epsilon_{rr} + \epsilon_{\theta\theta} =$ Transformed 1st invariant of strain

SOLUTIONS FOR DISPLACEMENT STRAIN AND STRESS COMPONENTS The solution for \bar{u} is obtained from (2) as

$$\bar{u} = \bar{c} r + \frac{D}{r}$$

(7)

63

DEF. Sci. J., Vol. 25, April 1975

where the constants \bar{C} and \bar{D} are determined from the boundary conditions given by

$$\bar{\sigma}_{rr}(a, w) = \bar{p}(w) \text{ or } \bar{p}, \ \bar{\sigma}_{rr}(b, w) = 0$$

'a' and 'b' are internal and external radius respectively. Using (8)

$$\bar{C} = -\left(\frac{a^2}{b^2 - a^2}\right) \left(1 - 2\bar{\nu}\right) \frac{\bar{p}}{2\bar{\mu}}$$
(9)

(8)

$$\bar{D} = -\left(\frac{a^2}{b^2 - a^2}\right) \frac{\bar{P}}{2\bar{\mu}}$$
(10)

On using (9) and (10), equations (3), (4), (5) and (6) yield solutions in transformed form

$$= -\left(\frac{a^2}{b^2-a^2}\right)\left(r+\frac{b^2}{r}\right)\left\{\left(1-2\bar{\nu}\right)\frac{\bar{p}}{2\bar{\mu}}\right\}$$
(11)

$$\bar{\epsilon}_{rr} = -\left(\frac{a^2}{b^2 - a^2}\right) \left(1 - \frac{b^2}{r^2}\right) \left\{ (1 - 2\bar{\nu}) \frac{\bar{p}}{2\bar{\mu}} \right\}$$
(12)

$$\bar{\epsilon}_{\theta\theta} = -\left(\frac{a^2}{b^2 - a^2}\right) \left(1 + \frac{b^2}{r^2}\right) \left\{ (1 - 2\bar{\nu}) \frac{\bar{p}}{2\bar{\mu}} \right\}$$
(13)

$$\bar{\sigma}_{rr} = -\left(\frac{a^2}{b^2 - a^2}\right) \left(1 - \frac{b^2}{r^2}\right) \bar{p}$$
(14)

$$\bar{\sigma}\theta\theta = -\left(\frac{a^2}{b^2 - a^2}\right)\left(1 + \frac{b^2}{r^2}\right)\bar{p}$$
(15)

$$\bar{\sigma}_{ss} = -\left(\frac{a^2}{b^2 - a^2}\right) 2\bar{\nu} \,\bar{p} \tag{16}$$

The solutions for an ebonite cylinder subject to constant internal pressure ' p_0 ' can be obtained in following manner,

$$\bar{p}(w) = \int_{0}^{\infty} p_0 H(t) e^{-iwt} dt = \left(\frac{p_0}{w}\right), \text{ where } H(t) = 1, t \ge 0 = 0 \text{ ; } t < 0$$
$$\bar{\mu} = \mu^* (iw) = K (iw)^{\nu}$$

determined by Lethersich and reported by Bland²

ü

 $\nu = \text{constant Poisson's ratio}$

By using the above relations and the Fourier inversion formula along with the table of Fourier transforms⁴, the time dependent solutions for displacement, strain and stress can be obtained as

$$u(r,t) = G_1\left(r + \frac{b^2}{r}\right)t^{r}$$
(17)

where

$$G_{1} = \left(\frac{a^{2}}{b^{2} - a^{2}}\right) \left(1 - 2\nu\right) \frac{p_{0}}{4\pi K} \sin \pi \left(1 + \nu\right) \left(-\nu\right)^{\frac{1}{2}}$$

- ν = Gamma function of $(-\nu)$

Choosing
$$\rho = \frac{r}{a}$$
, $R = \frac{b}{a}$, the displacement and strain components can be expressed in non-

dimensional form as

KANANGO & PATNAIK : Quasistatic Response of an Infinite Hollow Viscoelastic Cylinder

$$u^{*}(\rho, t) = \frac{u(r, t)}{C_{1} a} = \left(\rho + \frac{R^{2}}{\rho}\right) t^{\nu}$$
 (18)

Similarly strain components are

$$\boldsymbol{\epsilon_{rr}}^{*}\left(\boldsymbol{\rho},\,t\right) = \frac{\boldsymbol{\epsilon_{rr}}\left(r,\,t\right)}{C_{1}} = \left(1 - \frac{R^{2}}{\rho^{2}}\right)\,t^{\nu} \tag{19}$$

$$\epsilon_{\theta\theta}^{*}(\rho, t) = \frac{\epsilon_{\theta\theta}(r, t)}{C_{1}} = \left(1 + \frac{R^{2}}{\rho^{2}}\right) t^{\nu}$$
(20)

stress components are

$$\sigma_{rr}(r, t) = -\left(\frac{a^2}{b^2 - a^2}\right) \left(1 - \frac{b^2}{r^2}\right) p_0 H(t)$$
(21)

$$\sigma_{\theta\theta}(\mathbf{r},t) = -\left(\frac{a^2}{b^2 - a^2}\right) \left(1 + \frac{b^2}{\mathbf{r}^2}\right) \mathbf{p}_{\mathbf{0}} H(t)$$
(22)

$$\sigma_{zz}(r, t) = -\left(\frac{a^2}{b^2 - a^2}\right) 2 \nu p_0 H(t)$$
(23)

NUMERICAL COMPUTATIONS

The geometry of the cylinder is such that $\frac{o}{a} = R = 3$. $\nu = 0.014$. Since the response is sufficiently low with respect to time calculation of radial displacement $u^*(o, t)$, strain components $\epsilon^*_{--}(o, t)$ & $\epsilon^*\theta\theta(o, t)$

slow with respect to time calculation of radial displacement $u^*(\rho, t)$, strain components ϵ^*_n (ρ, t) & $\epsilon^*\theta\theta$ (ρ, t) are performed at different times between 10⁵ sec (approximately one day after loading) and 10⁷ sec. and at different radial locations. The calculated results are tabulated in Tables 1, and 2.

RADIAL DISPLACEMENT U^* CALCULATED AT DIFFERENT TIMES AND AT DIFFERENT RADIAL POINTS **ρ**⇒1 $\rho = 1.5$ $\rho = 2$ $\rho = 2.5$ $\rho = 3 \cdot 0$ i (seo) 8.2 105 11.177.6 5.384.675105.5 11.94 8.36 7.75 5.5 4.75 106 12.13 8.5 7.9 5.6 4.8510*** 8.65 8.05 12.32 5.67 $4 \cdot 925$ 107 12.52 8.76 8.15 5.00 5.76

TABLE 1

TABLE 2

STRAIN COMPONENTS, "" " " OO CALCULATED AT DIFFERENT TIMES AND AT DIFFERENT BADIAL POINTS

t (800)	ρ=1		<i>F</i>	ρ=2		ρ == 3	
	e* ₅₇	^{\$*} 00	€* ₽₽	¢* ₀₀	e*,,,	e*00	
10 ⁵ 10 ⁵⁻⁵ 10 ⁶ 10 ⁴⁻⁵ 10 ⁹	9·35 9·5 9·7 9·85 10·0	$ \begin{array}{r} 11 \cdot 17 \\ 11 \cdot 94 \\ 12 \cdot 13 \\ 12 \cdot 32 \\ 12 \cdot 52 \end{array} $	$-1 \cdot 46 \\ -1 \cdot 49 \\ -1 \cdot 51 \\ -1 \cdot 545 \\ -1 \cdot 562$	3.8 3.875 3.95 4.02 4.075	0 0 0 0 0	2.352.382.432.472.51	

In the present problem which relates to the viscoelastic medium, it is seen that even though the loading is time-independent the response is time-dependent. This can be attributed to the dependence of the properties of viscoelastic material on frequency on time. This time dependent strain or displacement is called the

65

creep. It is further noticed that the variation of strain component with time at a particular radial location for the viscoelastic cylinder under constant internal pressure hears an analogy to one suggested by Goldhoff* for C. -Mo-V steel at 538°C under constant stress.

ACKNOWLEDGEMENT

The authors are grateful to Prof. B. Mahapatra, Principal, University College of Engineering, Burla for encouragement.

REFERENCES

1. LEE, E.H., Quarterly of Applied Mathematics, 18 (1965), 183-190

and the second second

÷ . . .

1.50

13

2. BLAND, D.R., "The Theory of Linear Viscoelasticity" (Pergamon Press, N.w York) 1960, pp 76, 121-122.

1. **1**.

Sec. Spece

erren i de la la la la secta de la secta de la construction de la desta de la secta de la secta de la secta de

n an an the state of the state of

3. CHRISTENSEN, R.M., "Theory of Viscoelasticity: An Introduction" (Academic Press, New York and London), 1971, pp51, 307.

د بالای در این کرد برماند. افغا بالای با این کرد این این این کرد این این این ا

and the second second

and the second second

1 28 21

26.21

No Car

ويورد المدارية المؤسليكة الأوطن

Dail

 $\{\cdot\}$

-Sect

Q:Č

16 26 6 6 6 6

4 BATEMAN, H., "Tables of Integral Transforms" (McGraw Hill Book Company, New York), 1954.

5. GOLDHOFF, R.M., "Advances in Creep Design" (Applied Science Publishers Ltd., London), 1971, p 95.