# PIEZOMETRIC EFFICIENCY IN AN ORTHODOX GUN

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It has been shown that the muzzle velocity and the piezometric efficiency have greater values in case of a suitable moderated charge than in case of a single charge. The suitable moderated charge should consist of two components such that the pressure driving the shot remains absolutely constant throughout the period when the second component burns, the constant pressure being equal to the pressure at the burnt of the first component.

In a recent paper Ray<sup>1</sup> discussed the possibility of getting high piezometric efficiency in an orthodox gun by the use of a moderated charge. In fact in that paper it was demonstrated that if a moderated charge of two components burns in an orthodox gun the pressure during the period when the second component burns can be kept constant by a suitable choice of the size and shape of the second component of the moderated charge. Constancy of pressure gives a higher value to the piezometric efficiency but the amount of increment was not discussed in that paper. Moreover it will be of interest to know whether this increment in piezometric efficiency is associated with the increase of muzzle velocity. In the present paper it has been tried to throw some light on these points. In fact it has been shown that if firstly piezo metric efficiency and muzzle velocity is calculated with a charge of amount C, a size  $D/\beta$  and shape  $\theta$  and secondly piezometric efficiency and muzzle velocity is calculated with a moderated charge of two components having amounts  $C_1$  and  $C_2$  so that  $C = C_1 + C_2$ , the first component having size  $D_1/\beta_1 = D/\beta$  and shape  $\dot{\theta}_1 = \theta$  (same as the first case) and giving a constant pressure in the second stage of burning (by Ray's method). Then we have observed that piezometric efficiency and muzzle velocity are both greater in the second case. For simplicity in discussion we adopt an isothermal model although Ray considered his problem in the general non-isothermal model. Also Kapur<sup>3,4</sup> discussed the theory of moderated charges in the general non-isothermal model.

## BALLISTIC EQUATIONS FOR THE SINGLE CHARGE

The covolume term is entirely neglected and the kinetic energy term in the energy equation is neglected during the burning of the charge. The shot start pressure is taken to be zero. Then the equations are

$$FCZ = p\left(K_0 + Ax - \frac{C}{\delta}\right) \tag{1}$$

$$\omega_1 v \frac{dv}{dx} = Ap \tag{2}$$

$$Z = (1-f) (1+\theta f)$$
(3)

$$D \ \frac{df}{dt} = -\beta p \tag{4}$$

Making the following transformation

$$\xi = 1 + \frac{x}{l}$$

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$$Al = K_0 - \frac{C}{\delta}$$
$$\eta = \frac{vAD}{FC\beta}$$
$$= \frac{pAl}{FC} \text{ and } M = \frac{A^2D^2}{FC\beta^2\omega_1}$$

The above equations can be made dimensionless and the reduced equations are,

$$Z = \zeta \xi \tag{5}$$

$$M\zeta = \eta \; \frac{d\eta}{d\xi} \tag{6}$$

$$Z = (1 - f) \quad (1 + \theta f) \tag{7}$$

$$\xi = -\eta \, \frac{dg}{d\xi} \tag{8}$$

and

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The above equations were solved by the method of Crow given in H.M.S.O<sup>2</sup>. The solution gives,

$$f = 1 - \frac{\eta}{M} \tag{9}$$

$$\boldsymbol{\xi} = \left[\frac{(1+\theta)}{(1+\theta f)}\right]^{N} \tag{10}$$
$$N = \frac{M}{\theta}$$

where

 $\zeta = \frac{(1+\theta)^2 \, (\xi^{1/N} - 1)}{\theta \zeta^{1+2/N}} \tag{11}$ 

and

This expression for  $\zeta$  can be written as a function of  $\eta$  by (9) and (10). Again the maximum pressure occurs when

$$\eta = \frac{(1+\theta)}{\left(1+\frac{2\theta}{M}\right)} \tag{12}$$

and the condition for the true maximum is  $M > (1 - \theta)$  and the maximum pressure will occur before all burnt of the charge. If  $M \leq (1 - \theta)$  the maximum pressure will occur at all burnt.

Then the expression for the maximum pressure in non-dimensional form is given by

$$\zeta_{max} = \frac{(1+\theta)^2 (N+1)^{N+1}}{\theta (N+2)^{N+2}}$$
(13)

From the given transformation we have

$$p_{max} = \frac{FC}{Al} \frac{(1+\theta)^2 (N+1)^{N+1}}{\theta (N+2)^{N+2}}, \quad M > (1-\theta) \quad (14)$$

Again (13) for  $\theta = 0$  reduces to

$$p_{mass} = \frac{FC}{Al} = \frac{1}{e^M}, \quad (M > 1)$$

$$p_{mass} = \frac{FC}{Al} = \frac{1}{(1+\theta)^N}, \quad (M - 1 - \theta) \quad (15)$$

Also

Since the value of  $\zeta$  at f = 0, i.e. at all burnt is

$$\zeta_B = \frac{1}{(1+\theta)^N}$$

The value of  $\eta$  at all burnt is given by

$$\eta_B = M \text{ and } v_B = \frac{FC\beta}{AD} M$$
 (16).

The value of  $\xi$  at all burnt is given by

 $\xi_B = (1+\theta)^N$  and consequently the distance travelled by the shot at all burnt is  $x_B = l \left[ (1+\theta)^N - 1 \right]$ (17)

Now there are two possibilities :

(i) the shot may leave the muzzle before all kurnt of the charge, i.e.  $\frac{x_B}{d} > 1$  or

(ii) the shot may leave the muzzle after all burnt of the charge, i.e.  $\frac{x_B}{d} < 1, d$  being the length of the gun.

Again after all burnt the gaz expands adiabatically and the corresponding equations are

$$\omega v \frac{dv}{dx} = Ap \tag{18}$$

$$p \nabla^{\gamma} = \text{constant}$$
 (19.)

where V represents the volume. (19) can be written as,

$$pA^{\gamma} (x+l)^{\gamma} = p_{B} A^{\gamma} (x_{B} + l)^{\gamma}$$

$$p = p_{B} \left(\frac{x_{B} + l}{x+l}\right)^{\gamma}$$
(20)

From (18) and (20) we have,

$$v \frac{dv}{dx} = \frac{A}{\omega} p_{B} (x_{B} + l)^{\gamma} \frac{1}{(x+l)^{\gamma}}$$

Integrating one gets

$$v^{2} = v_{B}^{2} + \frac{2A}{\omega} \frac{p_{B}}{1-\gamma} (x_{B}+l)^{\gamma} \left[ (x+l)^{1-\gamma} - (x_{B}+l)^{1-\gamma} \right]$$

Since  $x = x_B$  when  $v = v_B$ 

Putting of the value of  $x_B$ ,  $v_B$  and  $p_B$  from (15) to (17) in the above formula for  $v^2$ , we have

$$w^{2} = \frac{F^{2}C^{2}\beta^{2}}{A^{2}D^{2}}M^{2} + \frac{2A}{\omega(1-\gamma)}\frac{FC}{Al(1+\theta)^{N}}\left\{l(1+\theta)^{N}\right\}^{\gamma} \times \left[(x+l)^{1-\gamma} - \left\{l(1+\theta)^{N}\right\}^{1-\gamma}\right]$$

If V be the muzzle velocity of the shot we have

$$V^{2} = \frac{F^{2}C^{2}\beta^{2}}{A^{2}D^{2}} M^{2} + \frac{2FC}{\omega(1-\gamma) \{l(1+\theta)^{N}\}^{1-\gamma}} \times \left[ (d+l)^{1-\gamma} - \{l(1+\theta)^{N}\}^{1-\gamma} \right]$$

The simplified equation for the muzzle velocity is given by

$$V^{2} = \frac{FC}{\omega_{1}} \left[ M + \frac{2}{1-\gamma} \left\{ \left( \frac{1+\frac{d}{l}}{(1+\theta)^{N}} \right)^{1-\gamma} - 1 \right\} \right]$$
(21)

If  $\bar{p}$  be the mean pressure which applied to the shot base for the total shot travel would give the observed muzzle velocity, then the corresponding equation connecting  $\bar{p}$  is

 $\frac{1}{2}\omega V^2 = Ad\bar{p}$ 

$$\bar{p} = \frac{\omega \bar{V}^2}{2 A d}$$
(22)

putting the value of  $V^2$  from (21) we have

$$\overline{p} = \frac{FC}{2Ad} \left[ M + \frac{2}{1-\gamma} \left\{ M + \frac{2}{1-\gamma} \left\{ M + \frac{2}{1-\gamma} \left\{ \left( \frac{1+\frac{d}{l}}{(1+\theta)^N} \right)^{1-\gamma} - 1 \right\} \right\} \right]$$
(23)

The piezometric efficiency is defined to be the ratio of the mean pressure to the maximum pressure.

Hence from (23) and (14) when

$$M > (1 - \theta)$$

Piezometric efficiency =  $\frac{P}{m}$ 

or

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$$=\frac{l}{2d}\frac{\theta (N+2)^{N+2}}{(1+\theta)^2 (N+1)^{N+1}} \left[M+\frac{2}{1-\gamma}\left\{\left(\frac{1+\frac{d}{l}}{(1+\theta)^N}\right)^{1-\gamma}-1\right\}\right]$$
for  $\left(\frac{x_B}{d}<1\right)$  (24)

and when  $M \leq (1 - \theta)$  from (23) and (15)

Piezometric efficiency = 
$$\frac{l}{2d} (1+\theta)^N \left[ M + \frac{2}{1-\gamma} \left\{ \left( \frac{1+\frac{1}{l}}{(1+\theta)^N} \right)^{1-\gamma} - 1 \right\} \right]$$
 (25)

Also the muzzle velocity in non-dimensional form is denoted by  $\eta_m$  which from (21) and the transformation formula reduces to

$$\eta_m^2 = M \left[ M + \frac{2}{1-\gamma} \left\{ \left( \frac{1+\frac{\alpha}{l}}{(1+\theta)^N} \right)^{1-\gamma} - 1 \right\} \right]$$
(26)  
for  $\left( \frac{x_B}{d} < 1 \right)$ 

All these formulae for piezometric efficiency and muzzle velocity will hold good if  $\frac{x_B}{d} < 1$ 

but if  $\frac{x_B}{d} > 1$ , i.e. the shot leaves the muzzle before all burnt of the charge, then from (10) for x = d the value of f is to be calculated and then  $\eta_m = M(1-f)$  gives the corresponding muzzle velocity. The mean pressure is then determined. For the calculation of maximum pressure in this case, the cases we have considered that f corresponding to ejection of the shot is less than f corresponding to  $p_{max}$  under the condition  $M > (1 - \theta)$  and therefore the maximum pressure was calculated by (14). In other cases the value of maximum pressure can be obtained numerically from the ballistic equations. The piezometric efficiency can be calculated from the definition. The numerical values of the muzzle velocity and piezometric efficiency are given for M = 1 and M = 2 for different values of l/d and  $\theta$ .

MODERATED CHARGE OF TWO COMPONENTS

Ballistic Equations when the First Component Burns

Under the same conditions as in the single charge, the equations become

$$F_1C_1Z_1 = p \left[ K_0 + Ax - \frac{C_1}{\delta_1} - \frac{C_2}{\delta_2} \right]$$
 (27)

$$\omega_1 v \quad \frac{dv}{dx} = Ap \tag{28}$$

$$D_1 \quad \frac{df_1}{dt} = -\beta_1 p \tag{29}$$

$$Z_1 = (1 - f_1) \ (1 + \theta_1 f_1) \tag{30}$$

These equations are to be solved with the initial conditions

x = 0, v = 0, p = 0 at  $f_1 = 1$  i.e.  $Z_1 = 0$ 

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We make the following usual substitution

$$\begin{split} \xi &= 1 + \frac{x}{l_1} \\ Al_1 &= K_0 - \frac{C_1}{\delta_1} - \frac{C_2}{\delta_2} \\ \eta &= v \frac{AD_1}{F_1 C_1 \beta_1} \\ \zeta &= p \frac{Al_1}{F_1 C_1} \\ M_1 &= \frac{A^2 D_1^2}{F_1 C_1 \beta_2 m_1} \end{split}$$

and

The above equations reduce to

$$Z_{1} = \zeta \xi$$

$$M_{1}\zeta = \eta \frac{d\eta}{d\xi}$$

$$\zeta = -\eta \frac{df_{1}}{d\xi}$$

$$Z_{1} = (1 - f_{1}) (1 + \theta_{1} f_{1})$$

$$(31)$$

Solving the above equations as in the case of single charge we find that the maximum pressure occurs when

$$\eta_1 = \frac{N_1 (1 + \theta_1)}{N_1 + 2}$$
 when  $N_1 = \frac{M_1}{\theta_1}$ 

and the condition for true maximum is

 $M_1 > (1 - \theta_1)$ 

If  $M_1 \leq (1 - \theta_1)$ , the maximum pressure occurs at all burnt of the component charge. Then the value of the maximum pressure is given by

$$p_{max} = \frac{F_1 C_1}{A l_1} \frac{(1+\theta_1)^2 (N_1+1)^{N_1} + 1}{\theta_1 (N_1+2)^{N_1} + 2}$$
(32)

The value of  $\zeta$  at  $f_1 \doteq 0$ , i.e. at burnt of the first component charge is given  $\zeta_{B1}$  where

$$\zeta_{B1} = \frac{1}{(1+\theta_1)^{N_1}}$$
(33)

Hence the value of the pressure  $p \operatorname{at} f_1 = 0$  is  $p_{R1}$  where

$$p_{B1} = \frac{F_1 C_1}{A l_1} \frac{1}{(1+\theta_1)^{N_1}}$$
(34)

Also  $x_{B1}$  the value of x where the first component charge burns out and  $v_{B1}$  the corresponding value of the velocity, then

$$\bar{v_{B1}} = -\frac{F_1 C_1 \beta_1}{A D_1} M_1$$
 (35)

$$x_{B1} = l_1 \left[ (1 + \theta_1)^{N_1} - 1 \right]$$
(36)

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Ballistic Equations when the Second Component Burns

The equations are :

$$F_1C_1 + F_2C_2Z_2 = p \left[K_0 + Ax - \frac{C_1}{\delta_1} - \frac{C_2}{\delta_2}\right] \qquad (37)$$

$$w_1 \quad \frac{dv}{dt} = Ap \tag{38}$$

$$D_2 - \frac{df_2}{dt} = -\beta_2 p \tag{39}$$

$$Z_2 = (1 - f_2) \ (1 + \theta_2 f_2) \tag{40}$$

We are to obtain the solution of these equations with initial conditions  $x = x_{B1}$ ,  $v = v_{B1}$  at  $f_2 = 1$ .  $x_{B1}$ ,  $v_{B1}$ ,  $p_{B1}$  are the value. of x, v, p when the first component just burns out.

Let us assume that the solution of the above equation is possible with

$$p = p_{B1} \tag{41}$$

and seek conditions so that the solutions may give  $x = x_{B1}$ ,  $v = v_{B1}$  at  $f_2 = 1$  and the systems of (37) to (40) may remain consistent for the solution  $p = p_{B1}$ .

 $\frac{dv}{dt} = - \frac{AD_2}{2}$ 

From (38) and (39)

or

$$\frac{d}{df_2} \left[ \frac{df_2}{dt}, \frac{dx}{df_2} \right] = - \frac{AD_2}{\beta_2 \omega_1}$$

which by (39) and (41) reduces to

$$\frac{d^2x}{df_2^2} = \frac{AD_2^2}{\beta_2^2 \omega_1 p_{B1}}.$$
 (43)

From (42)

$$v = \frac{F_1 C_1 \beta_1}{A D_1} M_1 + \frac{A D_2}{\beta_2 \omega_1} (1 - f_2)$$

$$v = v_{B1} \text{ at } f_2 = 1$$
(44)

since

Then the value of  $v \operatorname{at} f_2 = 0$  when the second component charge burns out

$$v_{B2} = \frac{F_1 C_1 \beta_1}{A D_1} M_1 + \frac{A D_2}{\beta_2 \omega_1}$$
(45)

From (37) and (40)

$$F_1C_1 + F_2C_2 (1 - f_2) (1 + \theta_2 f_2) = p_{B1} \left\{ K_0 + Ax - \frac{C_1}{\delta_1} - \frac{C_2}{\delta_2} \right\} (46)$$

(42)

Now we impose the condition that (37) gives  $x = x_{B1}$ ,  $v = v_{B1}$  at  $f_2 = 1$  and further (37) is consistent with (43).

Now  $x = x_{B1}$  will satisfy (46) if

$$F_1 C_1 = p_{B1} \left\{ K_0 + A x_{B1} - \frac{C_1}{\delta_1} - \frac{C_2}{\delta_2} \right\}$$
(47)

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which is true for (47) is obtained from (27) by considering burnt values. Now differentiating (37) and (40) w. r. t.  $f_2$ 

$$F_2C_2 \quad \frac{dZ_2}{df_2} = p_{B1} A \quad \frac{\frac{dW}{dt}}{\frac{df_2}{dt}}$$

$$\frac{dZ_2}{df_2} = (1 - f_2) \theta_2 - (1 + \theta_2 f_2)$$

and

and these equation with (39) yield

$$F_2C_2\left[ (1-f_2) \ \theta_2 - (1+\theta_2f_2) \right] = - \frac{AD_2}{\beta_2} \ \frac{dx}{dt}$$

Now  $v = v_{B1}$  and  $f_2 = 1$  will satisfy the above equation if

 $2\theta_2 =$ 

$$F_2C_2 (1 + \theta_2) = \frac{AD_2}{\beta_2} v_{B1}$$

Putting the value of  $v_{B1}$  the above equation reduces to

$$(1+\theta_2) = \frac{F_1 C_1}{F_2 C_2} \frac{D_2}{D_1} \frac{\beta_1}{\beta_2} M_1 \qquad (48)$$

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(49)

Now to satisfy the condition that (37) and (41) will be consistent, we differentiate (37) twice w. r. t.  $f_2$ .

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$$F_2C_2 \quad \frac{d}{df_2^2} = A p_{B1} \quad \frac{d}{df_2^2}$$

$$F_2C_2 \quad \frac{d^2Z}{df_2^2} = p_{B1} \quad \frac{A^2D_2^2}{\beta_2^2\omega_1p_{B1}}$$

$$\frac{d^2Z_2}{df_2^2} = -2\theta_2$$

Since

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we have

The simultaneous satisfaction of the equations (48) and (49) gives the condition that 
$$p=p_B$$
 may be a solution of the equations (37) to (40). From (49) it is evident that the second component must have progressively increasing burning surface.  
Introducing the dimensionless constant.

 $\frac{A^2D_2{}^2}{\beta_2{}^2\omega_1F_2C_2}$ 

$$\frac{F_2C_2}{F_1C_1} = \beta_0 \quad \text{and} \quad \frac{\frac{D_2}{\beta_2}}{\frac{D_1}{\beta_1}} = \alpha_0 \ .$$

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We have

and

$$1 + \theta_2 = \frac{\alpha_0}{\beta_0} M_1 \left\{ 2 \theta_2 = -\frac{M_1 \alpha_0^2}{\beta_0} \right\}$$
(50)

Eliminating  $\theta_2$  we have,

$$M_{1\alpha_{0}^{2}} + 2 M_{1\alpha_{0}} - 2\beta_{0} = 0.$$

$$\alpha_{0} = -1 + \sqrt{1 + \frac{2\beta_{0}}{M_{1}}}$$
(51)

giving

Since  $\alpha_0$  is always positive.

Now after all burnt the gas expands adiabatically and the corresponding equations are

$$\omega_1 v \quad \frac{dv}{dx} = Ap. \tag{52}$$

$$pV^{\gamma} = \text{constant}$$
 (53)

Equation (53) gives,

$$pA^{\gamma} (x + l_{1})^{\gamma} = p_{B1} A^{\gamma} (x_{B2} + l_{1})^{\gamma}$$

$$p = p_{B1} \left( \frac{x_{B2} + l_{1}}{x + l_{1}} \right)^{\gamma}$$
(54)

or

where  $x_{B2}$  is the distance traversed by the shot at all burnt of the second component, which is to be calculated.

Now from (52) to (54)

$$vdv = -\frac{A}{\omega_1} p_{B1} (x_{B2} + l_1) \frac{dx}{(x+l_1)^{\gamma}}$$

Integrating with the condition that  $x = x_{B2}$ ,  $v = v_{B2}$  one gets

$$v^{2} = v_{B2}^{2} + \frac{2Ap_{B1}}{\omega_{1}(1-\gamma)} (x_{B2}+l_{1})^{\gamma} \left[ (x+l_{1})^{1-\gamma} - (x_{B2}+l_{1})^{1-\gamma} \right]$$
(55)

Let us calculate the shot travel when the second component burns From (44), (39) and (41) we have

$$\frac{dx}{df_2} \quad \frac{\beta_2 p_{B1}}{D_2} = \frac{F_1 C_1 \beta_1}{A D_1} \quad M_1 + \frac{A D_2}{\beta_2 \omega_1} \quad (1 - f_2)$$

Integrating with the condition that  $x = x_{B1}$  at  $f_2 = 1$ ,

$$\frac{\beta_2 p_{B1}}{D_2} \left[ x - l_1 \left\{ (1 + \theta_1)^{N_1} - 1 \right\} \right] = M_1 \frac{F_1 C_1 \beta_1}{A D_1} (1 - f_2) + \frac{A D_2}{2\beta_2 \omega_1} (1 - f_2)^2$$

Let  $x = x_{B2}$  when the second component burns out, i.e. at  $f_2 = 0$ , then

$$x_{B2} = l_1 \left(1 + \theta_1\right)^{N_1} \left[1 + M_1 \alpha_0 + \frac{M_1 \alpha_0^2}{2}\right] - l_1$$

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$$+ M_0 \alpha_0 + \frac{M_1 \alpha_0^2}{2} = M_0 \tag{56}$$

Then

$$\boldsymbol{x_{B2}} = l_1 \left( 1 + \theta_1 \right)^{N_1} M_0 - l_1 \tag{57}$$

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From (45), (55) and (57) by putting the value of  $x_{B2}$  and  $v_{B2}$  we have

$$v^{2} = \left(\frac{F_{1}C_{1}\beta_{1}}{AD_{1}}M_{1} + \frac{AD_{2}}{\beta_{2}\omega_{1}}\right)^{2} + \frac{2A}{(1-\gamma)\omega_{1}}\frac{F_{1}C_{1}}{Al_{1}}\frac{1}{(1+\theta_{1})^{N_{1}}}l_{1}(1+\theta_{1})^{N_{1}}M_{0} \times \left[(x+l_{1})^{1-\gamma} - \left\{l_{1}(1+\theta_{1})^{N_{1}}M_{0}\right\}^{1-\gamma}\right]$$

If V be the muzzle velocity of the shot,

$$V^{2} = \frac{A^{2} D_{1}^{2}}{\beta_{1}^{2} \omega_{1}^{2}} \left[ \left(1 + \alpha_{0}\right)^{2} + \frac{2 M_{0}}{(1 - \gamma) M_{1}} \left\{ \left(\frac{1 + \overline{l_{1}}}{M_{0} (1 + \theta_{1}) N_{1}^{*}}\right)^{1 - \gamma} - 1 \right\} \right]$$
(58)

The mean pressure  $\overline{p}$  is given by  $\overline{p} = \frac{\omega_1 V^2}{2 A d}$ 

$$= \frac{\omega_1}{2Ad} - \frac{A^2 D_1^2}{\beta_1^2 \omega_1^2} \left[ (1+\alpha_0)^2 + \frac{2M_0}{(1-\gamma)M_1} \left\{ \left( \frac{(1+\overline{l_1})}{(1+\theta_1)^{N_1}M_0} \right)^{1-\gamma} - 1 \right\} \right]$$

Considering the expression for the maximum pressure for the two cases

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and (i) 
$$M_1 > (1 - \theta_1)$$
  
(ii)  $M_1 \leqslant (1 - \theta_1)$ 

we have the expression for the piezometric efficiency of the gun.

When

$$M_1 > (1 - \theta_1)$$

Piezometric efficiency =

$$\frac{l_{1}}{2 d} \frac{\theta_{1} (N_{1} + 2)^{N_{1} + 2}}{(1 + \theta_{1})^{2} (N_{1} + 1)^{N_{1} + 2}} \left[ M_{1} (1 + \alpha_{0})^{2} + \frac{2M_{0}}{1 - \gamma} \right] \cdot \left\{ \left( \frac{1 + \frac{d}{l_{1}}}{M_{0} (1 + \theta_{1})^{N_{1}}} \right)^{1 - \gamma} - 1 \right\} \right] \quad (59)$$

$$\left( \frac{x_{B1}}{d}, \frac{x_{B2}}{d} < 1 \right) \\
M_{1} \leq (1 - \theta_{1})$$

and when

for

for

Piezometric efficiency =

$$\frac{l_1}{2d} (1+\theta_1)^{N_1} \left[ M_1 (1+\alpha_0)^2 + \frac{2M_0}{1-\gamma} \left\{ \left( \frac{1+l_1}{M_0 (1+\theta_1)^{N_1}} \right)^{1-\gamma} - 1 \right\} \right] \\ \left( \frac{x_{B1}}{d} , \frac{x_{B2}}{d} < 1 \right)$$
(60)

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Let

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The corresponding expressions for the muzzle velocity in non dimensional form can be written as,

$$\eta_{m_3}^2 = M_1 \left[ M_1 \left( 1 + \alpha_0 \right)^2 + \frac{2M_0}{1 - \gamma} \left\{ \left( \frac{1 + \overline{l_1}}{M_0 \left( 1 + \theta_1 \right)^{N_1}} \right)^{1 - \gamma} - 1 \right\} \right] \quad (61)$$

Now for the moderated charge there are three possibilities:

(i)  $\frac{x_{B1}}{d} > 1$ , i.e. shot leaves the gun before all burnt of the first component such that

the constant pressure phase has not been reached. We are not concerned with this because we are to consider the cases under constant pressure phase with the second component.

(ii)  $\frac{x_{B1}}{d} < 1$  but  $\frac{x_{B2}}{d} > 1$ , i.e. the shot leaves the gun when the second component is burning

is burning.

(iii)  $\frac{x_{B1}}{d} < 1$  and  $\frac{x_{B2}}{d} < 1$ , i.e. the shot leaves the gun after all burnt of the second

component. We are concerned with the last two possibilities. In this case the muzzle velocity and piezometric efficiency are given by (59) to (61). For the second possibility the calculations of the muzzle velocity and piezo metric efficiency are as follows. From (46) and (50) for  $x = d, f_2$  have been calculated and with this value of  $f_2$  the muzzle velocity can be calculated from (44) and in non-dimensional form it reduces to

$$\eta_{m_2} = M_1 \left[ 1 + \alpha_0 \left( 1 - f_2 \right) \right]$$
(62)

The maximum pressure has been calculated on the same principle as in the case of single charge and hence the piezometric efficiency can also be calculated by deducing the mean pressure from the muzzle velocity and taking ratio of the mean pressure to the maximum pressure.

### NUMERICAL CALCULATION

Now we are going to establish numerically that both the muzzle velocity and piezometric efficiency have greater values for the moderated charges. For the simplicity of calculation we have considered that the propellant for the single charge and moderated charges have the same chemical composition, i.e.  $F = F_1 = F_2$ ,  $\delta = \delta_1 = \delta_2$  and  $\gamma = \gamma_1 = \gamma_2$ . Also we have stated that the ballistic of the single charge is equal to that of the first component, i.e.  $D/\beta = D_1/\beta_1$  and  $C = C_1 + C_2$ . Under these condition the ratio  $M/M_1$  reduces to  $C_1/C$  and we choose the value of M and  $C_1/C$ , the value of  $M_1$  has been determined. l/d and  $l_1/d$  are considered to be equal as we considered same gun in two cases. Also  $\beta_0 = F_2 C_2/F_1 C_1 = (C/C_1 - 1)$ . Hence when we choose  $C_1/C$ ,  $\beta$  can be found out and  $\alpha_0$ is then determined by (51).  $\frac{l}{d}=0.5, \theta=0, M=1$ 

Single	charge	- 10 - 10 - 10 - 10 - 10 - 10 - 10 - 10	100	Moderated charges					
M.V.	P.E.		C1/C (%)	M.V.	<b>P.E</b> .		0,/C (%)	M.V.	P.E.
1.093	0.8114	6	90 92 95	$1.12 \\ 1.105 \\ 1.1$	0.8222 0.8200 0.8190		97 98	1.098 1.095	0.8163 0.8157

 $\frac{l}{d} = 0.5, \theta = 1, M = 1$ 

Single o	harge		Moderated charges						<i>*</i>	
М.V.	P.E.		C <sub>1</sub> /C (%)	M.V.	P.E.		C1/C (%)	М.V.	P.E.	
1.44	0.7522	100	70	1.552	0.8422	9 S	88	1.437	0.803	
	- 114	10.00	72	1.531	0.8405	a section of a	90	- 1.413	0-796	
180 5		a na gasa	.78	1.511	0.8235		95	1.375	1.774	
			80	1.488	0.8224		98	1:353	0.763	
	2. 0		85	1.438	0.8136	10 M 10	1.25	v - 17 (17 (17 (17 (17		

$$\frac{l}{d}=0.2, \theta=0, M=0$$

, i	Single	charge	- A - 114		e <del>d</del> e la	Modera	ted cha	rges		
	M.V.	P.E.		(%)	M.V.	P.E.	e she	C./C (%)	M.V.	P <b>.E</b> .
,	1.562	0.6632		55	1.862	0.9157 -		80	1.766	0.8480
	10 10 400 C	1 A 1		60	1.840	0-9106	10.40	85	1.727	0.8100
		*.		65	1.824	0.8975		90	1.665	0.7529
				70	1.815	0.8950		95	1.616	0.7095
				75	1.796	0.8778		98	1.584	0.6827

ı		0.0	•					
T	-	0.2,	U	==	1,	M	=	1

Singl	e charge	신 국민 및 홈							
M.V.	P.E.		0,/0 (%)	М.V.	<b>P.E</b> .	14,15	C1/C (%)	M.V.	<b>P.E</b> .
1.709	0.4927	run el 1	45	2.532	0.8354		75	2.135	0.6776
	1. 1. 1. 1.		50	2.503	0.8224		80	2.049	0.6500
	2		55	2.501	0.8156		85	1.958	0.6015
			58	2.443	0.8114	1.04.53	90	1.851	0.5472
			60	2.395	0.7829	1.	95	1.784	0.5182
			65	2.301	0.7429		98	1.740	0.5051
	- E.	(c) = (c)	70	2.236	0.7100	1.2			0 0001

 $\frac{1}{d} = 0.1, \theta = 0, M = 1$ 

Sing	le charge			Moderated charges				
м.с.	<b>P.E.</b>	C1/C (%)	<b>M.V.</b>	P.E.	C <sub>1</sub> /C M.V. (%)	Р.Е.		
1.833	0.3207	60 80	$2 \cdot 488 \\ 2 \cdot 181$	0·8414 0·6457	90 2.068	0.5378		

Y		×.	$\frac{l}{d} = 0.1, \theta$	= 0, M = 2
Sing	gle charge			Moderated charges
M.V.	Р.Е.	Naria de Constante Santa Santa de Constante Santa d	C <sub>1</sub> /C M.V. (%)	P.E. C <sub>1</sub> /C M.V. P.E. (%)
2.348	· 0·7496		90 2.63	5 0·9295 98 2·610 0·9173

 $\frac{l}{d} = 0.1, \theta = 1, M = 2$ 

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•••••• ·	Single cl	narge				Moderated	charges	- 6	
N	<b>4.</b> V.	P.E.	- 	C <sub>1</sub> /C (%)	M.V.	Р.Е.	C <sub>1</sub> /C (%)	M.V.	P.E.
2	2.751	0.4483		85 87	$2.870 \\ 2.815$	0·7293 0·7025		$2 \cdot 792 \\ 2 \cdot 770$	0.6224 0.6223

 $\frac{l}{d}=0.1, \ heta=-0.1, \ M=1$ 

Single o	harge	Moderated charges	
M.V.	<b>P.E.</b>	$C_{1}/C$ M.V. P.E. $C_{1}/C$ M.V. (%)	P.E.
1.812	0.4709	$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	0.7986 0.6785
		$\begin{array}{cccccccccccccccccccccccccccccccccccc$	$0.6385 \\ 0.6345$

 $\frac{l}{d} = 0.1, \ \theta = -0.2, \ M = 1$ 

Single charge		- 1. 	M	oderated charges	
M.V. P.E.	-	C1/C (%)	M.V. ' P.E.	C <sub>1</sub> /C (%)	M.V. P.E.
1·787 0·4800	- 	80 90	$\begin{array}{cccc} 1 \cdot 932 & 0 \cdot 6925 \\ 1 \cdot 872 & 0 \cdot 5430 \end{array}$		1 · 859 0 · 5315

 $\frac{l}{d} = 0.1, \ \theta = -0.6, \ M = 1$ 

Single ch	arge			Moderated	charges		
<b>M.V.</b>	P.E.		<i>C</i> <sub>1</sub> / <i>C</i> M.V. (%)	P.E.	C1/C (%)	M.V.	·P.E.
 1.611	0.5905	5	98 1.627 95 1.644	0.6115 0.6400	90 87	$1 \cdot 692 \\ 1 \cdot 712$	$0.6735 \\ 0.6852$

 $\frac{l}{d}=0.1, \ \theta=-0.8, \ M=1$ 

Single charge			Modera	ted cha	rges		
<b>M.V.</b> P.E.	C <sub>1</sub> /O (%)	M.V.	P.E.		C1/O (%)	M.V.	P.E.
1.345 0.6656	 98 95	1·316 1·311	0.6610 0.6750	-	90 87	$1 \cdot 282 \\ 1 \cdot 123$	0.6820 0.6937

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